

# Elevated Temperature Mechanical Properties of Al Alloy AA6063/SiC<sub>p</sub> MMCs

Tarek A. Khalifa and Tamer S. Mahmoud

**Abstract**—Mechanical properties at elevated temperatures of as-cast and extruded AA6063 aluminum alloy reinforced with SiC particles (SiC<sub>p</sub>) were studied. Hot extrusion was carried out with reduction in area of 25% on the as cast composites which were prepared by the vortex method. Tensile tests carried out at room temperature showed that for the as-cast composites, the addition of SiC<sub>p</sub> up to 10% by weight improves the strength but reduces ductility. Further addition of SiC<sub>p</sub> reduces both the strength and the ductility of the composites. At 150 and 300 °C the matrix alloy exhibits higher strength than the composites, but the extruded composites have better strength than as-cast composites. Time rupture creep tests carried out at 300 °C showed that the composites exhibits higher creep resistance as compared to the matrix alloy except at relatively low stresses where the matrix has a better creep resistance.

**Index Terms**—Mechanical Properties, Creep, Metal Matrix Composites, Extrusion.

## I. INTRODUCTION

Aluminum alloys reinforced with particulates are more attractive than traditional aluminum alloys for applications requiring higher stiffness and strength [1,2]. Reinforcement by particles or short fibers of SiC has proved to be especially advantageous since it offers composite materials having virtually isotropic properties at low cost. Recently, there has been a growing interest in using metal matrix composites for high temperature applications, especially under creep conditions, in aircraft and automobile engines technologies.

Although many studies have been devoted to the characterization of the room-temperature mechanical behaviour of aluminum metal matrix composites, only limited high temperature data are available [3-7]. The influence of addition of ceramic particles to aluminum alloys on the elevated temperature strength is difficult to predict. For example, Nieh et al. have studied the tensile properties of composites reinforced with SiC particles over a range of temperatures from room temperature up to 400 °C [6]. They showed that the composites exhibits better mechanical strength up to 150 °C as compared to the unreinforced alloy, but at higher temperatures the mechanical properties of the composites is almost the same as the material without SiC particles addition. Luster et al. [4] showed that there is no beneficial strengthening at elevated temperatures of Al<sub>2</sub>O<sub>3</sub>

reinforced 6061Al composites, but the strength was actually reduced in comparison with the unreinforced alloy. Several studies [8-11] have shown that particle reinforcement improves the creep resistance of aluminum alloys. In the range 150 to 250 °C, the creep behaviour of particle reinforced aluminum alloys is dependent on the matrix strength and matrix strengthening mechanisms. In matrices with strong and stable microstructures; precipitation strengthened or solution strengthened, only modest creep strengthening from the addition of reinforcement is observed. In weak matrices such as pure aluminum, the presence of reinforcing particles alters the matrix dislocation structure during creep, resulting in significant matrix strengthening.

Stir casting route is the most promising one for synthesizing discontinuous reinforcement dispersed aluminum alloy matrix composites because of its relative simplicity and easy adaptability with all shape casting processes [12]. The composites produced by stir casting have many defects such as particles clustering and high porosity content, which have a deleterious effect on the mechanical properties [13-14]. These defects can be altered by mechanical working of the composites such as extrusion or rolling [15]. Mechanical working of monolithic alloy, usually carried out above the recrystallization temperature, not only deforms the alloy into the desired shapes and sizes but also refines the microstructure and reduces the porosity content, which in turn improves the mechanical properties. However, this is not completely true in the case of composite materials due to incompatible deformation characteristics of the matrix and the ceramic reinforcement particles. This incompatibility leads to changes in the optimum processing conditions.

The objective of the present work is to study the effect of extrusion on elevated temperatures tensile and creep behaviour of aluminum metal matrix composites produced by the vortex method.

## II. EXPERIMENTAL PROCEDURES

The aluminum alloy used as the matrix was 6063 having the composition given in Table 1. The composites were prepared by the vortex method. SiC ceramic particles (SiC<sub>p</sub>) having a size range of 30 to 150 μm and an average size of 60 μm were used for reinforcement. The matrix was reinforced with 5, 10, and 15 % wt of SiC<sub>p</sub>.

The vortex method involved melting the matrix by heating it to a temperature of 710 °C in an electric resistance furnace. After the aluminum was melted, it was degassed with dry nitrogen gas to minimize the oxidation of the molten metals. A simple steel stirrer attached to a variable speed motor was used to stir the melt. The stirring speed was 750 rpm and the stirring time was about 10-15 minutes. Before introducing the ceramic particles to the melt, they were properly preheated to a temperature of 300 °C for 2 hours. After the

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Tarek A. Khalifa is a Professor at Mechanical Engineering Department, Shoubra Faculty of Engineering, Benha University, Cairo, Egypt. (Phone: 002-010-1343329; fax:002-02-22023336; e-mail: tkhalifa50@yahoo.com).

Tamer S. Mahmoud is an Assistant Professor at Mechanical Engineering Department, Shoubra Faculty of Engineering, Benha University, Cairo, Egypt. (Phone: 002-018-3657294; fax: 002-02-22023336; e-mail: tsamir@Benha-univ.edu.eg).

stirring process was completed, the crucible was taken from the furnace and the composite was poured into a metallic mold. After the as-cast composites were prepared, they were cut and machined into round bars having a diameter of 17 mm for extrusion. Extrusion was carried out on the as-cast composites with a reduction ratio of 25% in area. The billet was heated to a temperature of 500 °C for half an hour before the composite was extruded. Both as-cast and extruded composites were solution heat treated at 530±3 °C for three hours and then quenched in cold water. After cooling, specimens were cut and artificially aged at 175±1 °C to determine the peak hardness level attainable during age hardening and the time required for achieving such a hardness level. Vickers hardness test measurements were carried out using a load of 10 kg.

Table 1. Chemical composition of AA6063 matrix alloy

Si	Fe	Cu	Mg	Ti	Al
0.38	0.15	0.007	0.46	0.11	Balance

Tensile and creep tests were carried out to determine the effect of adding SiC<sub>p</sub> with different weight percentages and the effect of deformation on the mechanical properties of AA6063 after heat treatment. Tensile tests were done at room temperature, 150, and 300 °C. Tensile specimens were prepared from the as-cast and the extruded composites. Tensile specimens having a diameter of 3 mm and a gauge length of 16 mm were machined longitudinally from the extruded billets. Tests were done using an Avery-Dennison universal testing machine with a load capacity of 100 KN and a gripping device for threaded-end specimen according to ASTM standard A 370-74 G 25. For each condition, three specimens were tested. After each test, the ultimate tensile strength ( $\sigma_{UTS}$ ), the 0.2% offset yield strength ( $\sigma_{YS}$ ), and the ductility  $\delta\%$  (measured by % elongation) were calculated.

Conventional high temperature creep rupture tests were carried out at 300 °C, in which a single tensile load is applied in a continuous fashion throughout each test until rupture occurs. The temperature was measured with three thermocouples at the top, middle and bottom of the specimens and the difference between the three thermocouples readings were not allowed to exceed ± 3 °C. The strain was measured using an extensometer having a range ±2 mm. Three different stress levels were used in the creep tests which are 10, 20, and 40 MPa. The rupture time was recorded and the rupture ductility was calculated. Fig. 1 shows the dimensions of the creep specimen used in the present investigation.

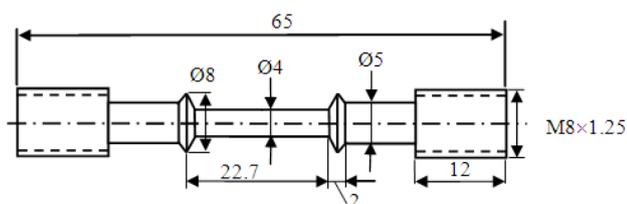


Fig.1. The Creep specimen

The microstructure of each tested condition was examined in both longitudinal and transverse directions of extrusion using optical microscopy. The actual volume fraction of particles was determined using quantitative metallographic techniques. Particle agglomeration was measured using line

intercept method. The porosity of the composites was measured using the typical Archimedes (water displacement) method. Scanning electron microscopy (SEM) was used to study the fracture surfaces after tensile and creep tests.

### III. RESULTS AND DISCUSSION

Examination of microstructure showed that the matrix grain size of the as-cast AA6063 was in the range of 120 to 170  $\mu\text{m}$  with an average value of 145  $\mu\text{m}$ , while, the grain size of the extruded alloy ranged from 50 to 100  $\mu\text{m}$  with average value of 75  $\mu\text{m}$  this is clearly noticed in Fig. 2a,b. The most apparent difference in metallographic features between the as-cast and extruded composites is that SiC particles, are present as banded clusters in the as-cast composites, and become more uniformly distributed in the extruded composites in both longitudinal and transverse directions as shown in Fig. 2 (c, d and e).

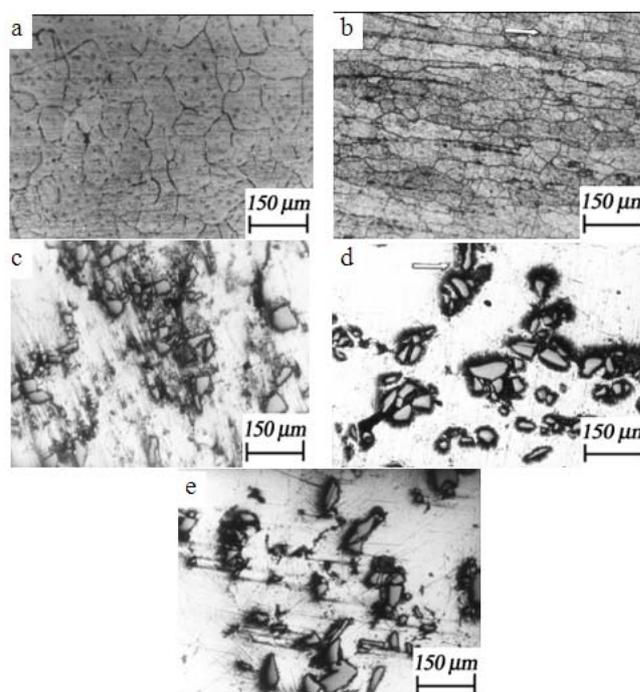


Fig.2. Effect of extrusion on the microstructure of composites and matrix alloy (a) as-cast matrix (b) extruded matrix (c) as-cast 15%wt SiC<sub>p</sub> composite (d) extruded 15%wt SiC<sub>p</sub> composite – longitudinal direction (the arrow indicates the extrusion direction) (e) extruded 15%wt SiC<sub>p</sub> composite – transverse direction.

Particle agglomeration was measured as volume fraction using line intercept method. Fig. 3 shows the volume fraction of agglomeration and agglomeration size versus the weight % of SiC<sub>p</sub> for the as-cast and extruded composites. For the as-cast composites the agglomeration volume fraction increased with increasing weight % of SiC<sub>p</sub> as shown in Fig. 3a. Also, Fig. 3b shows that the extruded composites have reduced agglomeration size when compared to the as-cast composites. This may be due to the shear flow of the aluminum matrix and grain splitting phenomena, which is found to be more extensive in the composites as compared with the unreinforced matrix [9].

The porosity of the composites before and after extrusion was measured. Fig. 4 shows the percentage volume fraction of porosity of as-cast and extruded composites. The figures

show that with increasing the weight percentage of  $\text{SiC}_p$ , the porosity of the composites increases. This may be due to the increase of stirring time required to disperse the  $\text{SiC}_p$  which increases the air bubbles entering the slurry [12]. The extruded composites exhibit a lower porosity content as compared with the as-cast samples, this is due to the compressive strains involved in the extrusion process.

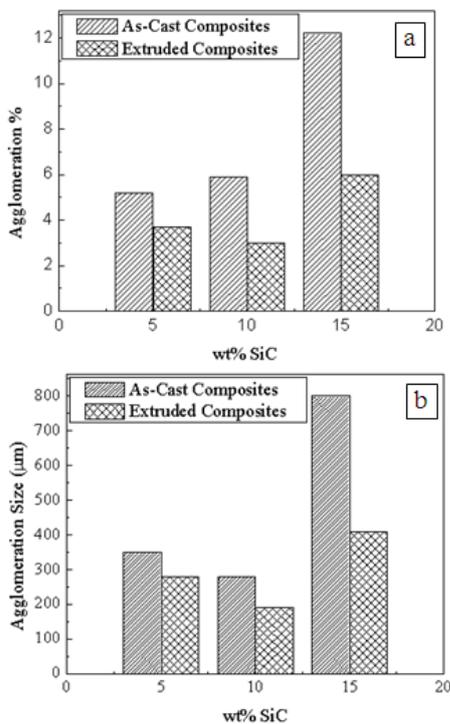


Fig.3. Effect of extrusion on (a) the agglomeration % and (b) the agglomeration size for composites with different weight percentages of  $\text{SiC}_p$ .

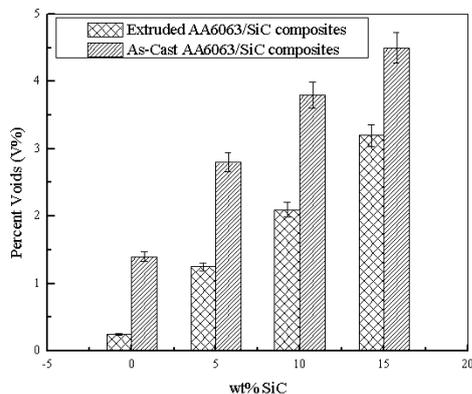


Fig.4. A plot showing the variation of voids % for both extruded and as-cast composites with various weight percentages of  $\text{SiC}_p$ .

Figure 5 shows the age hardening curves for the as-cast and extruded composites. It was found that in all conditions, the addition of  $\text{SiC}_p$  to the matrix reduces the time required to reach the peak hardness as compared to the matrix alloy (3-6 hours for composites and 8-9 hours for matrix alloy). Also, increasing the weight % of  $\text{SiC}_p$  reduces the time required to reach the peak hardness. Both as-cast and extruded composites showed the same aging behaviour. It was found that addition of  $\text{SiC}_p$  to the matrix alloy increased the hardness and this increase was maintained during the

different stages of artificial aging. The highest hardness value of 69 VHN was observed for the as-cast composites having 15% wt  $\text{SiC}_p$ . The accelerated aging behaviour observed in the present investigation may be a result of an increased dislocation density in the vicinity of  $\text{SiC}_p$ , which is due to a large difference in the coefficient of thermal expansion between  $\text{SiC}_p$  and the matrix. The higher dislocation density can both aid the diffusion of solute atoms and serve as nucleation sites, thereby leading to a more rapid precipitation process [16-17].

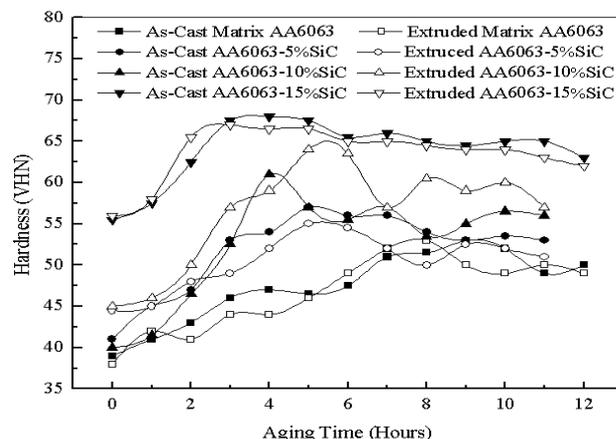


Fig.5. Aging curves for composites (as-cast and extruded).

Figure 6 shows the tensile properties of the matrix alloy and the composites (as-cast and extruded) at ambient, 150 and 300 °C. The room temperature tensile test results show that for the as-cast composites there is a slight increase in tensile strength with increasing the wt % of  $\text{SiC}_p$  up to a value of about 10%. Further increase in wt % of  $\text{SiC}_p$  resulted in a decrease in tensile strength. A similar trend is observed for the extruded composites. This may be due to the high porosity content, and clustering of particles in the composites. The 0.2 % offset yield strength of the as-cast composites showed a more pronounced increase with increasing weight percent of  $\text{SiC}_p$  up to a value of 10% which is again followed by a noticeable decrease with further increase of the weight percent of  $\text{SiC}_p$  to a value of 15 %. The values of the 0.2 % offset yield strength of the extruded composites followed a similar trend showing a less pronounced increase with addition of  $\text{SiC}_p$ . The extruded composites were generally stronger than the as-cast composites.

The tensile tests carried out at 150 °C showed that, for the as-cast conditions, increasing the wt % of  $\text{SiC}_p$  reduced both the tensile strength and 0.2 % offset yield strength as compared to the unreinforced alloy. For extruded conditions the ultimate tensile strength (UTS) increased with addition of  $\text{SiC}_p$  up to 10 wt %. Further increase in wt % of  $\text{SiC}_p$  is accompanied by a decrease in strength. The values of the 0.2 % offset yield strength followed a similar trend but with the peak value occurring at 5% wt of  $\text{SiC}_p$ . The tensile tests carried out at 300 °C indicated that the addition of  $\text{SiC}_p$  to the matrix decreases both the tensile strength and the 0.2 % offset yield strength. Generally, extrusion raised the strength. Increasing the weight % of  $\text{SiC}_p$  causes a slight reduction in the tensile strength of the extruded conditions. A similar trend was observed for values of the 0.2 % offset yield strength. The reduction in strength of the composites may be due to that at 300 °C the primary  $\text{MgSi}_2$  precipitates coarsen

more rapidly due to the presence of SiC<sub>p</sub> and this results in a more rapid decrease in the matrix strength which is the dominant factor in the strength of the composites at this temperature.

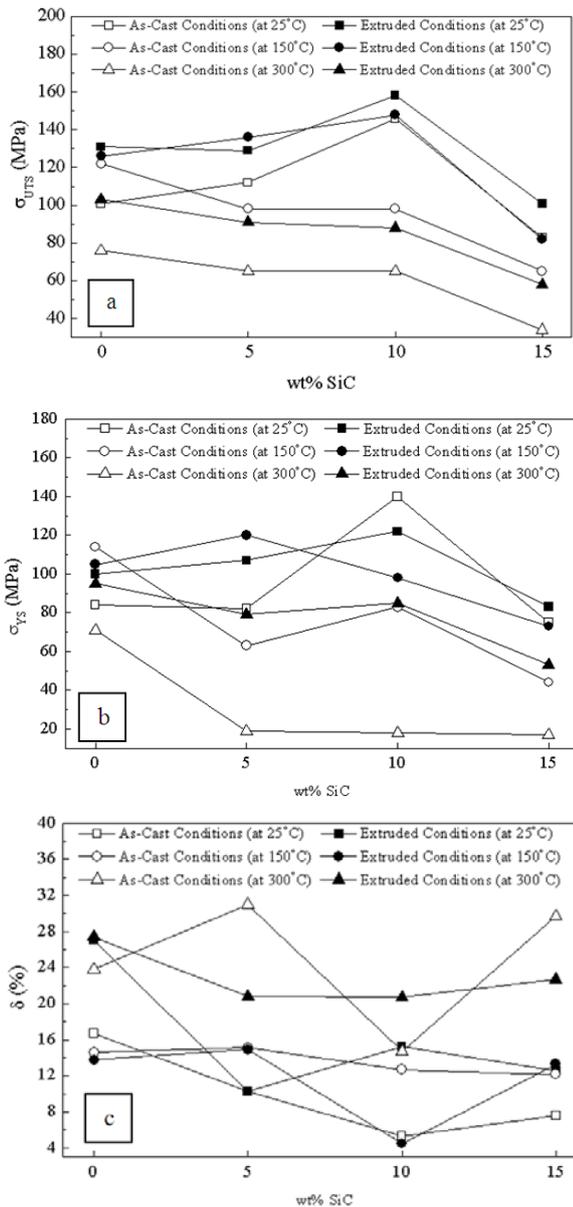


Fig.6. Variation of tensile properties with volume fraction of SiC<sub>p</sub> at different temperatures (a) tensile strength;  $\sigma_{UTS}$  (b) 0.2 % offset yield strength;  $\sigma_{YS}$  (c) elongation ;  $\delta$ %.

At room temperature increasing the wt % of SiC<sub>p</sub> reduces the ductility (measured as elongation %) of the composites as compared with the unreinforced alloy and the ductility of the extruded composites were higher than that of the as-cast composites. Extrusion, generally, raised the ductility of the matrix material at both room temperature and elevated temperature. These improvements achieved by extrusion may be due to reduction in matrix grain size, reduction of the porosity content, and the more uniform distribution of SiC<sub>p</sub>. For testing at elevated temperatures increasing the wt % of SiC<sub>p</sub> did not show a consistent effect on ductility and the effect of extrusion on ductility of the composites did not follow any clear trend.

Fig. 7 shows the rupture lifetimes of the matrix and composites obtained from creep tests. It is noticed that in

the as-cast conditions, the composites exhibits longer creep rupture time except at low stresses (10 MPa) where the matrix has longer creep life times. A similar trend was also observed for the extruded composites. No consistent effect on creep life time was observed with varying the weight percent of the reinforcing particles. An example of the time-strain creep curves is shown in Fig. 8 for the as-cast composites at a stress level of 10 MPa. Table 2 lists the values of the minimum creep rates for both the as-cast and extruded composites. The results revealed that in the as-cast conditions, the matrix has higher values of minimum creep rates as compared to the composites except at 20 MPa where the 5% wt SiC<sub>p</sub> composite has the maximum value. Also in the extruded conditions the matrix exhibits higher values of minimum creep rate as compared to the composites except at 40 MPa where the 10 % wt SiC<sub>p</sub> composite has the maximum value .

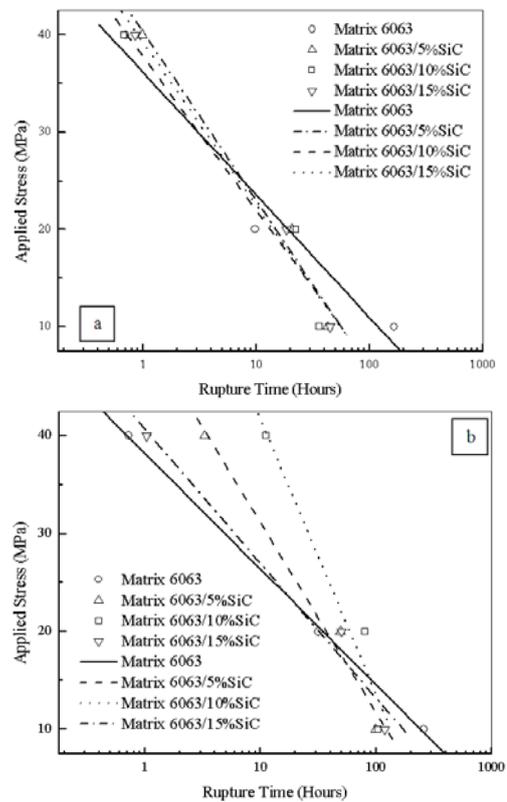


Fig.7. Creep rupture life times of (a) as cast and (b) extruded composites at 300 °C. Specimens were tested at three applied stresses 10, 20 and 40 MPa.

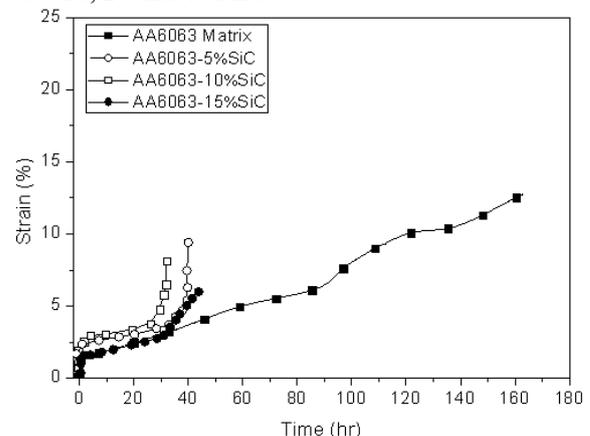


Fig. 8. Time-Strain creep curves for the as-cast alloys at 10 MPa.

Table 2. Minimum creep rate of the composites (%/hr).

Condition	Weight% SiC <sub>p</sub>	Stress (MPa)		
		10	20	40
As-Cast	0	0.1520	0.0830	8.000
	5	0.0070	0.2860	1.7800
	10	0.0012	0.0333	4.6000
	15	0.0026	0.1330	1.6000
Extruded	0	0.0330	0.3810	3.2000
	5	0.0010	0.1550	2.6000
	10	0.0025	0.1350	5.0000
	15	0.0019	0.2960	3.5000

Power law exponents have been established using minimum creep rates for both as-cast and extruded composites, using the following relation:-

$$\dot{\epsilon} = K\sigma^n \quad (1)$$

Where  $\dot{\epsilon}$  is the minimum creep rate,  $\sigma$  is the applied stress,  $K$  is a material constant, and  $n$  is the power law exponent which is also a material constant. Table 3 summarizes the values of the power law exponent ( $n$ ) and the constant  $K$  for the matrix and both as-cast and extruded composites. The calculated values of  $n$  show that, the composites have higher values of  $n$  as compared to the unreinforced alloys. This behavior is in agreement with previous work [8-11]. In the as-cast composites, increasing the weight percentage of the SiC<sub>p</sub> added to the matrix alloy increases  $n$  to reach its maximum value of 5.9 at 10 %wt SiC<sub>p</sub>. Any further increase in the SiC<sub>p</sub> weight percentage reduces the power law exponent. The power law exponent of the extruded composites has an approximately constant value range 5.4 and 5.6. Extrusion increases the values of the power law exponent except for the 10% wt SiC<sub>p</sub> composites where the as-cast composite has a slightly higher value of the exponent  $n$  as compared to the extruded composites.

Table 3. Values of the power law exponent ( $n$ ) and constant  $K$  for the investigated composites.

Weight % SiC <sub>p</sub>	As-cast composites		Extruded composites	
	$n$	$k$	$n$	$k$
0	2.85	$8.88 \times 10^{-5}$	3.3	$1.74 \times 10^{-5}$
5	4.87	$3.67 \times 10^{-8}$	5.6	$3.08 \times 10^{-9}$
10	5.9	$1.02 \times 10^{-9}$	5.4	$8.76 \times 10^{-9}$
15	4.63	$7.70 \times 10^{-8}$	5.4	$1.10 \times 10^{-8}$

The general feature associated with the fracture of the particle reinforced composite material can be summarized in terms of macroscopically brittle behaviour of the specimens, which fractured without any evidence of necking. Examination of the fracture surfaces of the failed tensile specimens at room temperature revealed the mechanism of the failure in the composites. Fig. 9a shows the fracture of the unreinforced as-cast specimen. It shows an even distribution of dimples, indicative of ductile failure. In composites the ductile dimples were still present, but now contain reinforced particles and the size of dimples was dependant on the size of

SiC<sub>p</sub>. It was noted that extrusion increases the interface strength between the particles and the matrix and decreases the porosity content. This is reflected by the smaller area % of sites indicative of particle decohesion in the fracture surface of the extruded composites. Particle pull-out was sometimes observed as shown in Fig. 9b. The pull-out particles were coated with aluminum matrix.

The fracture surfaces of the high temperature tensile specimens are quite different when compared to the room temperature fracture. As shown in Fig. 9c, an intergranular fracture mechanism has occurred in the matrix alloy although the fracture surface still contains dimples. The fracture surfaces of the composites show that decohesion along the interface between the SiC<sub>p</sub> and the matrix is still an integral fracture mechanism at high temperature, see Fig. 9d.

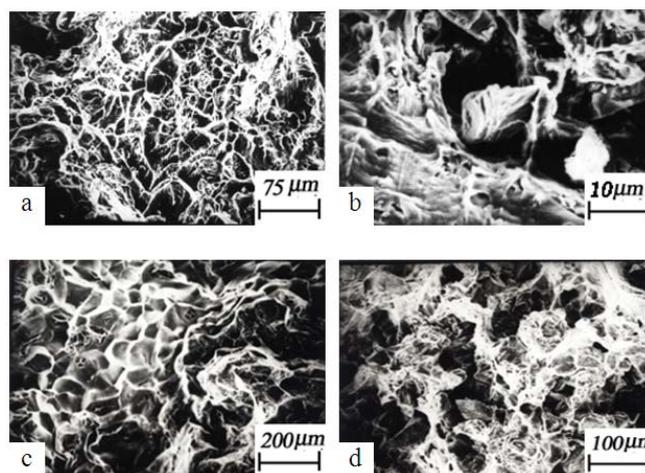


Fig.9. SEM micrographs of the tensile fracture surfaces of AA6063/SiC<sub>p</sub> composites; (a) room temperature as cast unreinforced; (b) room temperature extruded 10% SiC<sub>p</sub>; (c) At 300 °C, extruded unreinforced and (d) At 300 °C, extruded 10% SiC<sub>p</sub>.

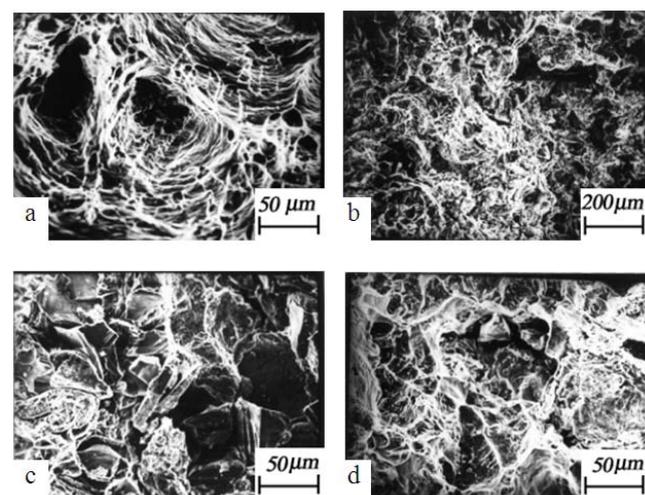


Fig.10. SEM micrographs of the creep fracture surfaces of AA6063/SiC<sub>p</sub> at 300 °C; (a) Extruded unreinforced at 10 MPa; (b) Extruded 10% SiC<sub>p</sub> at 20 MPa; (c) As cast 10% SiC<sub>p</sub> at 20 MPa and (d) As cast 5% SiC<sub>p</sub> at 40 MPa.

The creep fracture surface of some selected creep specimens are shown in Fig.10. Fracture surfaces appearance is essentially similar to those of tensile fracture at elevated temperatures. The fracture surface of the unreinforced alloy (as-cast and extruded) shows a ductile fracture with uniform

fracture surface as shown in Fig. 10a The fracture surfaces of the composites show that fracture occurs by void nucleation and growth around the grain boundaries of the matrix, while the reinforcing particles along the boundaries delaminate. The final fracture occurs by the link up of the voids at decohered particles leading to a dimpled fracture surface as shown in Fig. 10b. In addition to intergranular separation the fracture surfaces of the as-cast composites show extensive particles clustering as shown in Fig. 10c, this particles clustering is less observed on the fracture surfaces of the extruded composites. The final failure for reinforced alloys at high stresses is mostly intergranular in nature as shown in Fig. 10d with little evidence of reinforcement particle fracture. Some decohesion is apparent at the particles/matrix interface close to the final fracture surface.

#### IV. CONCLUSIONS

1. Extrusion with 25% reduction in area % reduces the porosity content of the as-cast composites and causes redistribution of  $\text{SiC}_p$  clusters, resulting in a more uniform distribution of the  $\text{SiC}_p$ .
2. Composites exhibit a significant acceleration in kinetics of precipitation in compare with the unreinforced matrix alloy. This acceleration is attributed to a decrease in the incubation time required to achieve the peak hardness.
3. The addition of  $\text{SiC}_p$  to AA6063 improves the strength (but reduces ductility) of the alloy at room temperature, up to 10 wt%, then the strength decreases with further increase in  $\text{SiC}_p$  content. At elevated temperature the results indicated that the addition of the  $\text{SiC}_p$  decrease the tensile strength as compared to the matrix alloy. Extruded composites showed a similar trend but with relatively higher values of strength.
4. The as-cast and extruded composites exhibit higher creep resistance as compared to the unreinforced alloy except at relatively low stresses below 10 MPa where the matrix shows a better creep resistance.
5. The as-cast and extruded matrix alloy has a power law exponent ( $n$ ) close to 3, while the as-cast composites have ( $n$ ) ranging between 4 and 6 which is higher than that of the matrix alloy. Power law exponent of the extruded composites is approximately constant and was in range 5.4 to 5.6, which is generally higher than that of the as-cast composites.
6. The matrix material generally exhibits a higher value of minimum creep rate as compared to the composites for both as cast and extruded conditions.

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